

PARAMETER IDENTIFICATION USING MODAL DATA

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PARAMETER IDENTIFICATION USING EXPERIMENTAL MODAL DATA

Introduction: Parameter Identification

- (1) Direct Approach - using actual sensor measurements to obtain parameters (e.g., ERA, ITD)
- (2) Indirect Approach - identifies parameters using measured modal data*

* in this work, only the natural frequencies are used to identify physical parameters

Equations of Motion

The motion of a distributed structure with zero damping is governed by a PDE of the form

$$\mathcal{L} u(P, t) + m(P) \ddot{u}(P, t) = f(P, t), \quad P \in D$$

$$B_i u(P, t) = 0, \quad P \in S \quad i = 1, 2, \dots, p$$

u = displacement at position P at time t

m = mass distribution

\mathcal{L} = stiffness differential operator of order $2p$

f = external force density

B_i = homogeneous differential operators of order ranging from zero to $2p-1$

ω

Discretization

Rayleigh-Ritz method: $u(P,t) = \tilde{\psi}^T(P) \tilde{q}(t)$

$\tilde{\psi}(P)$ - n-vector of admissible functions

$\tilde{q}(t)$ - n-vector of generalized coordinates

$$M \ddot{\tilde{q}}(t) + K \tilde{q}(t) = \tilde{Q}$$

$$M = \int_D m(P) \tilde{\psi}(P) \tilde{\psi}^T(P) dD, \quad K = \int_D \tilde{\psi}(P) \tilde{\psi}^T(P) dD$$

Discretization (cont'd)

Rayleigh-Ritz Type Parameter Expansion: $m(P) = \sum_{r=1}^g \alpha_r m_r(P)$

$$\mathcal{L} = \sum_{r=1}^h \beta_r \mathcal{L}_r$$

$$M = \sum_{r=1}^g \alpha_r M_r, \quad K = \sum_{r=1}^h \beta_r K_r$$

$$M_r = \int_D \tilde{\psi}(P) m_r(P) \tilde{\psi}^T(P) dD = \text{rth mass matrix}$$

$$K_r = \int_D \tilde{\psi}(P) \mathcal{L}_r \tilde{\psi}^T(P) dD = \text{rth stiffness matrix}$$

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Iterative Approach Using Least Squares

$$g(t) = \sum_{r=1}^n x_r e^{i\omega_r t}$$

$$\sum_r^2 M x_r = K x_r \quad (\text{Actual Model})$$

$$\sum_{0r}^2 M x_{0r} = K_0 x_{0r} \quad (\text{Postulated Model})$$

To refine the parameters α_{0r} , β_{0r} , we update them such that the theoretical natural frequencies converge to the measured natural frequencies of the actual structure.

$$\text{Postulated Parameters: } p_0 = [\alpha_{01} \alpha_{02} \dots \alpha_{0g} \beta_{01} \dots \beta_{0h}]^T$$

$$\text{Measured Natural Frequencies: } \tilde{\omega} = [\omega_1 \omega_2 \dots \omega_f]^T$$

$$\text{Postulated Natural Frequencies: } \tilde{\omega}_0(p_0) = [\omega_{01} \omega_{02} \dots \omega_{0f}]^T$$

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Iterative Approach (cont'd)

$$\text{Sensitivity Analysis: } \Delta \tilde{\omega} = \frac{\partial \tilde{\omega}}{\partial \tilde{p}} \Delta \tilde{p} \quad , \quad \left[\frac{\partial \tilde{\omega}}{\partial \tilde{p}} \right] = \left[\frac{\partial e_i}{\partial p_i} \right]$$

$$\Delta \tilde{\omega} = \tilde{\omega} - \tilde{\omega}(p_0), \quad \Delta \tilde{p} = \tilde{p} - p_0$$

\tilde{p} = updated parameters

Least-Squares Solution:

$$\Delta \tilde{p} = \left(\left[\frac{\partial \tilde{\omega}}{\partial \tilde{p}} \right]^T \left[\frac{\partial \tilde{\omega}}{\partial \tilde{p}} \right] - I \right)^{-1} \left[\frac{\partial \tilde{\omega}}{\partial \tilde{p}} \right]^T \Delta \tilde{\omega}$$

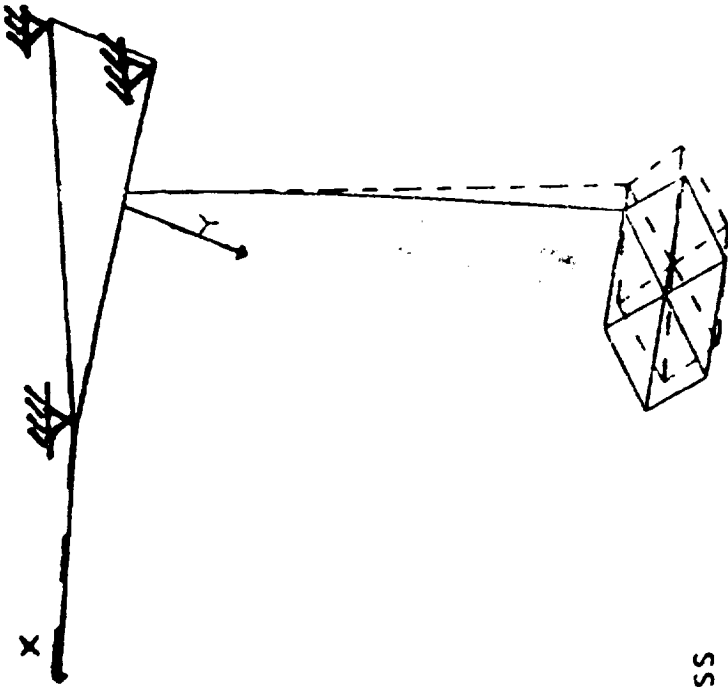
Iterative Approach (cont'd)

$$\text{Jacobian } \left(\frac{\partial w}{\partial p} \right): \frac{\partial w_r}{\partial p_1} = \frac{1}{\omega_r} \left(\frac{\partial}{\partial p_1} \left[\sum_{r=1}^T \frac{\partial K}{\partial p_1} \right] - \sum_{r=1}^T \frac{\partial M}{\partial p_1} \frac{x_r}{\omega_r} \right)$$

$$\frac{\partial M}{\partial p_1} = M_1, \quad \frac{\partial K}{\partial p_1} = 0 \quad (i = 1, 2, \dots, g)$$

$$\frac{\partial M}{\partial p_i} = 0, \quad \frac{\partial K}{\partial p_i} = K_{i-g} \quad (i = g+1, g+2, \dots, g+h)$$

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Numerical Example

SCALE MODEL:

Parameters:

m_A = antenna mass

I_{XA} = antenna moment of inertia with respect to roll axis

m = mast mass density

EI = mast bending rigidity

m_i = sensors and actuators at $z = -41.0, -87.8$ ($i = 1, 2$)

I_{x_i} = additional mass moment of inertia ($i = 1, 2$)

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Numerical Example

SCOLE MODEL

Kinetic Energy:

$$T(t) = \frac{1}{2} \int_0^L m \dot{w}^2(z,t) dz + \frac{1}{2} \sum_{i=1}^2 \{ [m_i \dot{w}^2(z_i,t) + I_{xi} [\dot{w}'(z_i,t)]^2] \\ + \frac{1}{2} m_A \dot{w}^2(L,t) + \frac{1}{2} I_{xA} [\dot{w}'(L,t)]^2 \}$$

Potential Energy:

$$V(t) = \frac{1}{2} \int_0^L \{ EI [w''(z,t)]^2 + P(z) [w'(z,t)]^2 \} dz$$

$w(z,t)$ - transverse displacement from equilibrium position

$P(z)$ = axial load

Numerical Example

- We used an $n = 4$ degree of freedom model:

$$M\ddot{q} + Kq = F$$

- Natural frequencies agree well with those obtained experimentally (data provided by Lee, Williams and Sparks)
- We used the first 2 natural frequencies to update the bending rigidity EI and the mass moment of inertia of the antenna.

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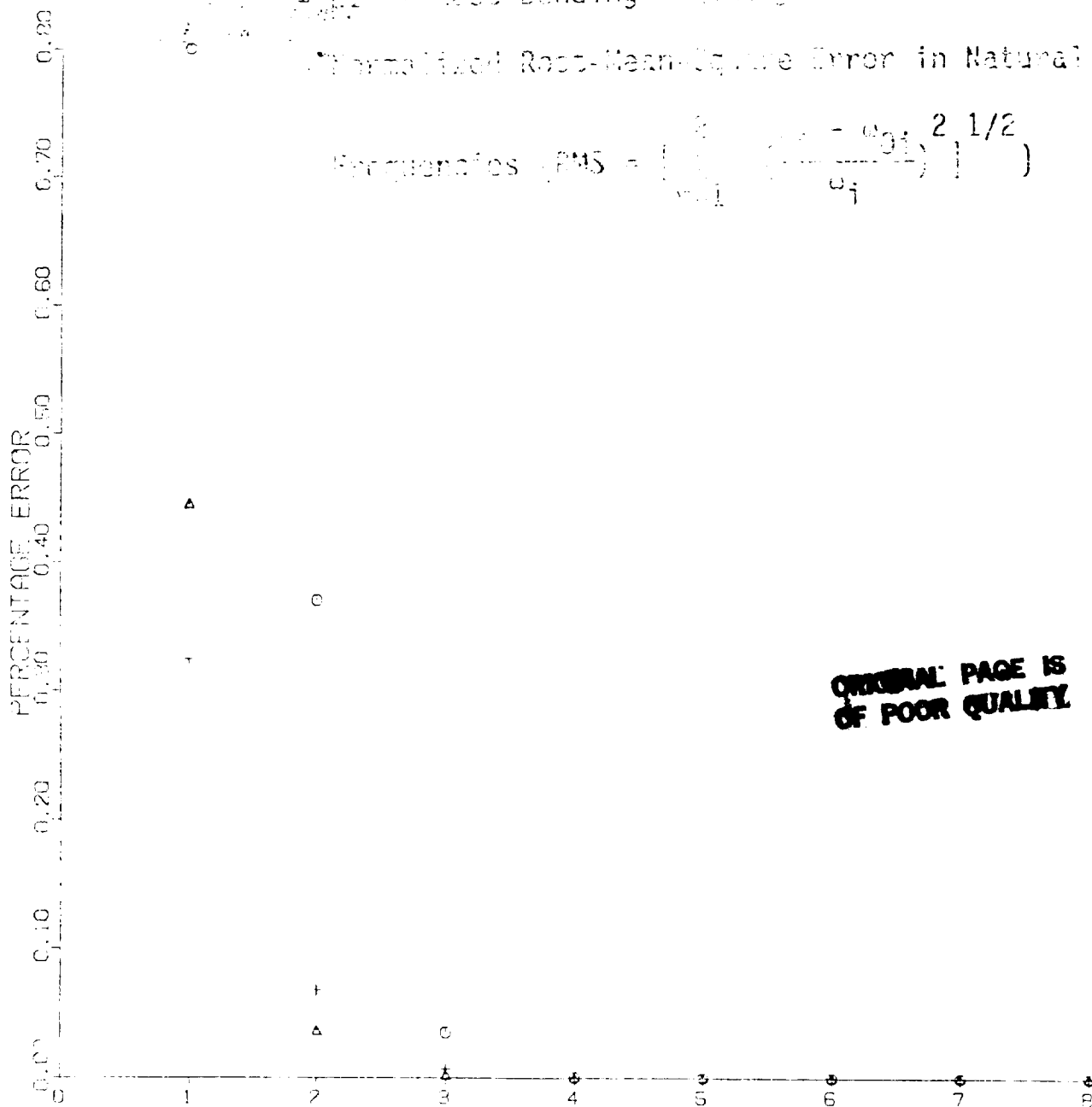
se I: Zero Error in Natural Frequencies

- I_{XA} Antenna Mass Moment of Inertia

- Δf Root Banding Frequency

Normalized Root-Mean-Square Error in Natural

$$\text{Frequencies (RMS)} = \left[\frac{1}{n} \sum_{i=1}^n \left(\frac{\omega_i - \omega_{0i}}{\omega_i} \right)^2 \right]^{1/2}$$



Case II: 1% Error in ω_1

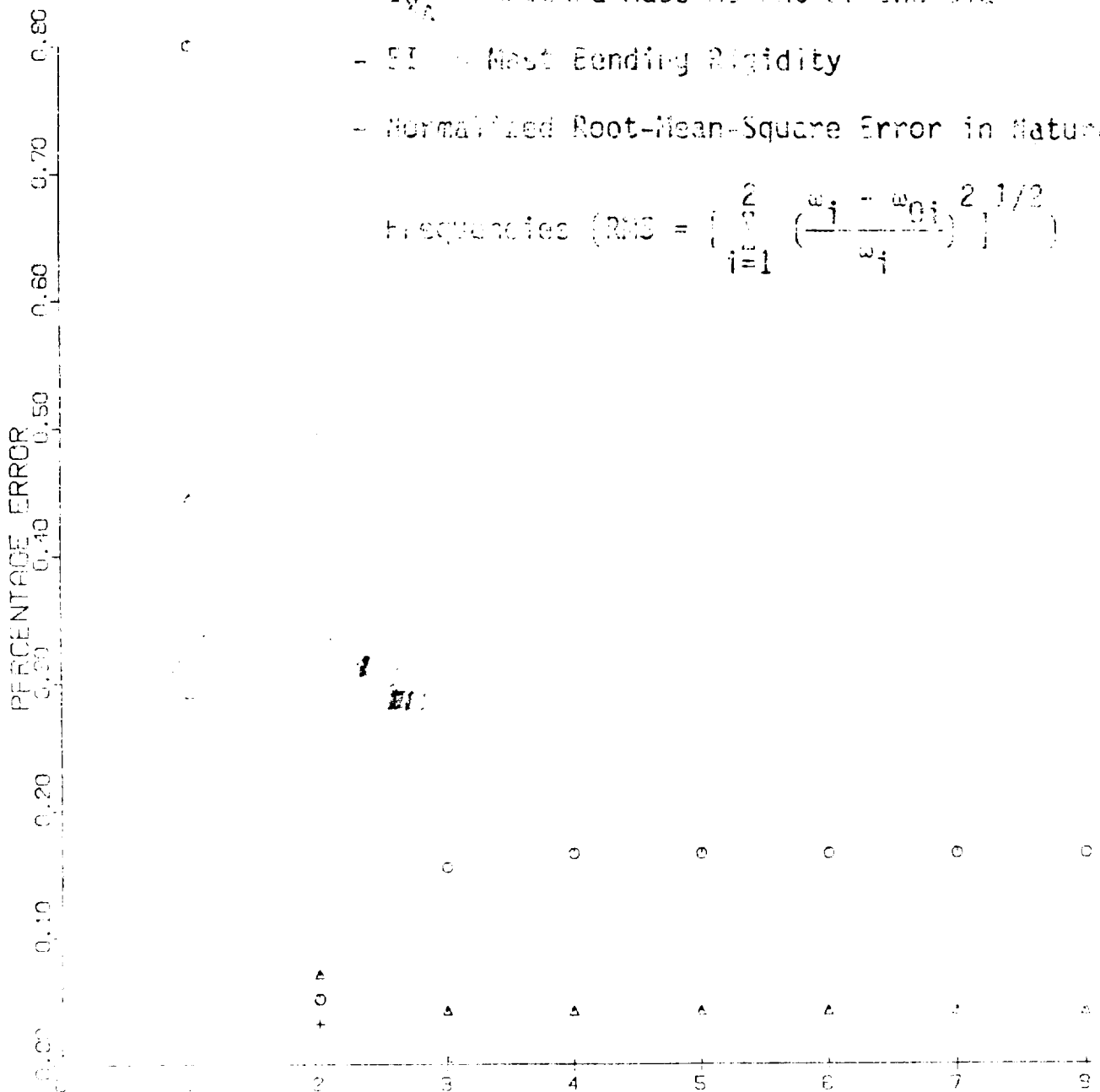
5% Error in ω_2

- I_{y_A} = Antenna Mass Moment of Inertia

- EI = Mast Bending Rigidity

- Normalized Root-Mean-Square Error in Natural

$$\text{Frequencies (RMS)} = \left[\frac{2}{n} \sum_{i=1}^n \left(\frac{\omega_i - \omega_{0i}}{\omega_i} \right)^2 \right]^{1/2}$$



Case III: 2% Error in ω_1

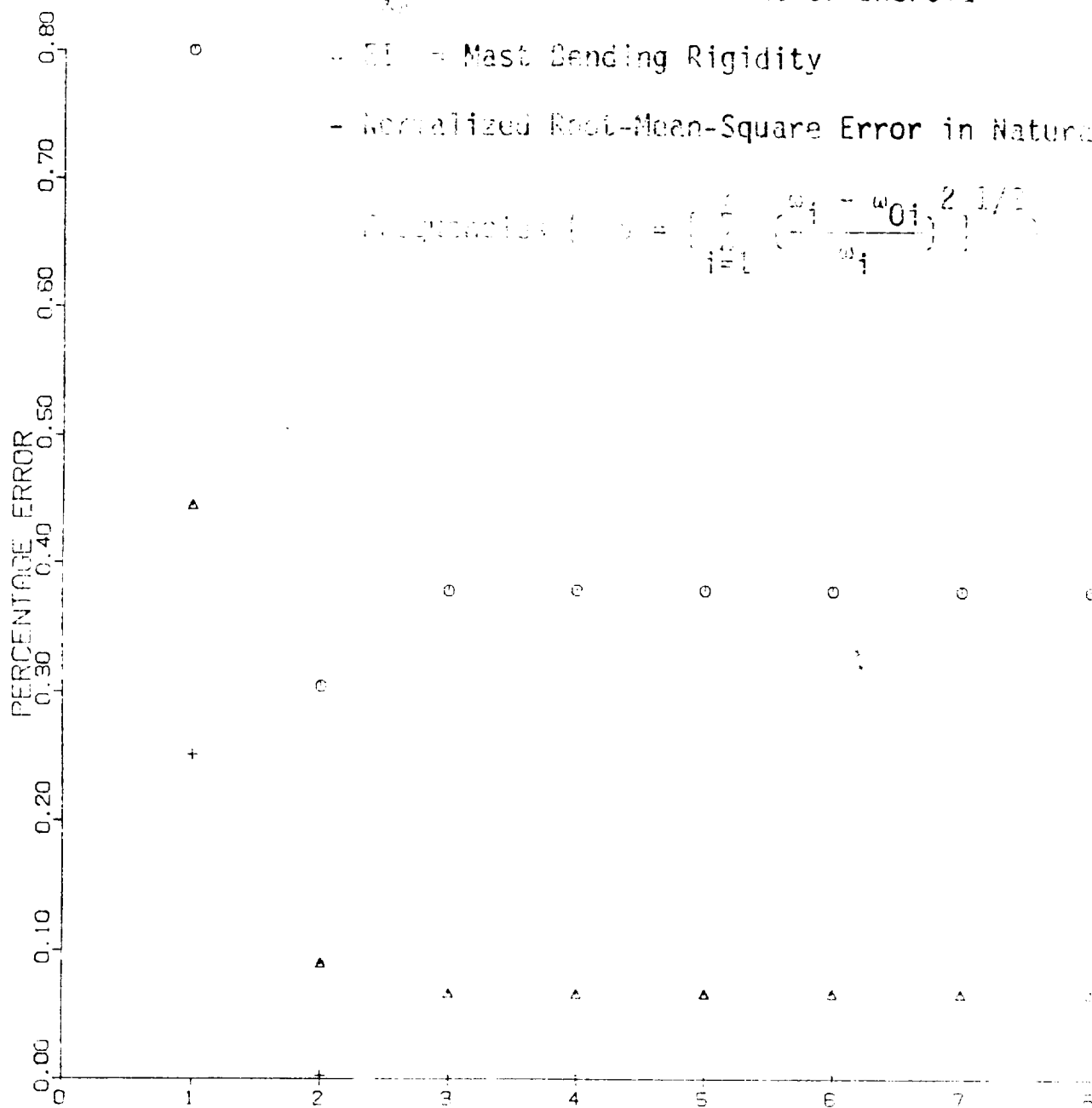
10% Error in ω_2

- I_{X_0} = Antenna Mass Moment of Inertia

- EI = Mast Bending Rigidity

- Normalized Root-Mean-Square Error in Natural

$$\text{Frequency } (\%) = \left(\frac{1}{N} \sum_{i=1}^N \left(\frac{\omega_i - \omega_{0i}}{\omega_i} \right)^2 \right)^{1/2}$$



Conclusions

- Indirect approach is used to obtain physical parameters
- Results indicate that the parameters converge quickly
- The algorithm is relatively insensitive to noise

